Flight test report

 Manufacturer
 Windtech Paragliders

 Address
 Francisco Rodríguez, 7

 3201 GIJON - Asturias
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 Spain, PO Box 269 33280
 None

 Type of glider
 Combat SC 23

 Trimmer
 not available

Certification number Date of flight test Place of test PG 109.2007 26/10/2007 Villeneuve



Classification D

Test Pilot Claude Thurnheer Harness Sky Axel II M 42cm Total weight in flight 80 kg Alain Zoller Sky Paragliders - Axel 2 M 90 kg

		Min weight	Max weight
1. Inflation/Ta			
	Rising behaviour	Overshoots, shall be slowed down to avoid front collapse	C Hangs back E
	Special take off technique required		A No A
2. Landing	Special landing technique required	No	A No A
3. Speed in st			,
	Trim speed more than 30 km/h	Yes	A Yes A
	Speed range using the controls larger than 10 km/h	Yes	A Yes A
	Minimum speed	25 km/h to 30 km/h	3 25 km/h to 30 km/h E
4. Control mo			
	Max. weight in flight up to 80 kg Symmetric control pressure/travel	Approximately appatent 40 cm to FE cm	C not available
	Max. weight in flight 80 kg to 100 kg	Approximately constant, 40 cm to 55 cm	10t available
	Symmetric control pressure/travel	not available	0 Approximately constant, 45 cm to 60 cm
	Max. weight in flight greater than 100 kg		
	Symmetric control pressure/travel	not available	0 not available
5. Pitch stabil	ity exiting accelerated flight		
	Dive forward angle on exit		A Dive forward less than 30° A
	Collapse occurs	No	A No A
6. Pitch stabil	ity operating controls during accelerated flight	Na	
7. Roll stabilit	Collapse occurs y and damping	No	A No A
	Oscillations	Reducing	A Reducing A
8. Stability in			
	Tendency to return to straight flight	Spontaneous exit	A Spontaneous exit A
9. Behaviour i	n a steeply banked turn		
	Sink rate after two turns	More than 14 m/s	B More than 14 m/s E
10. Symmetric	c front collapse	Desire heat meeter than 45%	Desking healt less than 45%
	Entry	0 0	C Rocking back less than 45° A D Spontaneous in less than 3 s A
	Recovery	further 3 s	
	Dive forward angle on exit		A Dive foward 0°to 30°, Keeping course A
	Cascade occurs		A No A
	With accelerator		
	Entry		C Rocking back less than 45° A
	Recovery	•	B Spontaneous in less than 3 s
	Dive forward angle on exit		A Dive foward 30°to 60°, Keeping course E
11 Exiting do	Cascade occurs ep stall (parachutal stall)	No	A No A
The Exiting de	Deep stall achieved	Yes	A Yes A
	Recovery		A Spontaneous in 3 s to 5 s
	Dive forward angle on exit		A Dive forward 30°to 60° E
	Change of course	Changing course less than 45°	A Changing course less than 45° A
	Cascade occurs	No	A No A
12. High angle	e of attack recovery		
	Recovery Cascade occurs		A Spontaneous in less than 3 s A A No A
13. Recovery	from a developed full stall		
y	Dive forward angle on exit	Dive forward 0°to 30°	A Dive forward 30°to 60° E
	Collapse		A No collapse A
	Cascade occurs (other than collapse)	No	A No A
	Rocking back		A Greater than 45°
44	Line tension	Most line tight	A Most line tight A
14. Asymmetr			
	With 50% collapse-Maximum dive forward or roll angle Change of course until re-inflation	Less than 90°, Dive or roll angle 15° to 45°	A Less than 90°, Dive or roll angle 15° to 45°
	Re-inflation behaviour		A Spontaneous re-inflation
	Total change of course	•	A Less than 360° A
	Collapse on the opposite side occurs		A No A
	Twist occurs		A No A
	Cascade occurs	No	A No A
	With 75% collapse-Maximum dive forward or roll angle		
	Change of course until re-inflation Re-inflation behaviour		C 90° to 180°, Dive or roll angle 45° to 60° C A Spontaneous re-inflation A
	Total change of course	•	A Spontaneous re-inflation A A Less than 360° A
	Collapse on the opposite side occurs		A Yes, causing turn reversal
	Twist occurs		A No A
	Cascade occurs		A No A
	With 50% collapse and accelerator-Maximum dive forward or		
	With 50% collapse and accelerator-Maximum dive forward or Change of course until re-inflation Re-inflation behaviour	Less than 90°, Dive or roll angle 15° to 45°	A 90° to 180°, Dive or roll angle 15° to 45° A Spontaneous re-inflation A

	Total change of course	Less than 360°	А	Less than 360°	А
	Collapse on the opposite side occurs	No	A	No	A
	Twist occurs	No	A	No	A
	Cascade occurs	No	A	No	A
	With 75% collapse and accelerator-Maximum dive forward of		~	110	~~~
	Change of course until re-inflation	90° to 180°, Dive or roll angle greater than 90°	D	90° to 180°, Dive or roll angle 45° to 60°	С
	Re-inflation behaviour	Spontaneous re-inflation	A	Spontaneous re-inflation	A
	Total change of course	Less than 360°	A	Less than 360°	A
	Collapse on the opposite side occurs	Yes, no turn reversal	C	Yes, no turn reversal	С
	Twist occurs	No	A	No	Α
	Cascade occurs	No	Α	No	Α
15. Direction	nal control with a maintained asymmetric collapse				
	Able to keep course	Yes	А	Yes	А
	180° turn away from the collapsed side possible in 10 s	Yes	Α	Yes	А
	Amount of control range between turn and stall or spin	More than 50 % of the symmetric control travel	Α	More than 50 % of the symmetric control travel	Α
16. Trim spe	eed spin tendency				
	Spin occurs	No	Α	No	Α
17. Low spe	ed spin tendency				
	Spin occurs	Yes	D	No	А
18. Recover	y from a developed spin				
	Spin rotation angle after release	Stops spinning in less than 90°	А	Stops spinning in less than 90°	А
	Cascade occurs	No	A	No	A
19. B-line st				110	~
TO: D IIIC St	Change of course before release	Change of course less than 45°	А	Change of course less than 45°	А
	Behaviour before release	Remains stable with straight span	Â	Remains stable with straight span	A
			B		
	Recovery	Spontaneous in 3 s to 5 s	-	Spontaneous in 3 s to 5 s	B
	Dive forward angle on exit	Dive forward 0° to 30°	A	Dive forward 30° to 60°	Α
	Cascade occurs	No	A	No	А
20. Big ears					
	Entry procedure	Standard technique	А	Dedicated controls	А
	Behaviour during big ears	Stable flight	А	Stable flight	А
	Recovery	Recovery through pilot action in less than a	В	Spontaneous in less than 3 s	А
		further 3 s			
	Dive forward angle on exit	Dive forward 0° to 30°	Α	Dive forward 0° to 30°	Α
21. Big ears	in accelerated flight				
	Entry procedure	Dedicated controls	Α	Dedicated controls	Α
	Behaviour during big ears	Stable flight	А	Stable flight	А
	Recovery	Recovery through pilot action in less than a	в	Recovery through pilot action in less than a	В
		further 3 s		further 3 s	
	Dive forward angle on exit	Dive forward 0° to 30°	А	Dive forward 0° to 30°	А
	Behaviour immediately after releasing the accelerator while	Stable flight	Â	Stable flight	Â
22 Bahavia	our exiting a steep spiral		A	otablo liight	A
ZZ. Denavio	Tendency to return to straight flight	Spontaneous exit	А	Spontaneous exit	А
		•		Less than 720°, spontaneous recovery	A
	Turn angle to recover normal flight	Less than 720°, spontaneous recovery	A		A
	Sink rate when evaluating spiral stability [m/s]	17 m/s		22 m/s	
23. Alternat	ive means of directional control	N N			
	180° turn achievable in 20 s	Yes	Α	Yes	Α
	Stall or spin occurs	No	A	No	A
24. Any othe	er flight procedure and/or configuration described in the us				
	Procedure works as described	not available		not available	0
	Procedure suitable for novice pilots	not available	0	not available	0
	Cascade occurs	not available	0	not available	0
Comments	of test pilot				
	Comments	no		The glider are a tendenc to stay in the back and	
				•	



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