Flight test report

Manufacturer Ozone Gliders Address 2, Queens Drive

ddress 2, Queens Drive LA46LN UK

Representive David Dagault
Type of glider Octane FLX 22
Trimmer not available

 Certification number
 PG 129.2008

 Date of flight test
 20/2/2008

 Place of test
 Villeneuve



Classification C

Test PilotSeiko FukuokaClaude ThurnheerHarnesssupair altiplumeGin Genie III M 45cmTotal weight in flight57 kg90 kg

		Min weight		May waight	
1. Inflation/Ta		win weight		Max weight	
	Rising behaviour	Smooth, easy and constant rising	Α	Smooth, easy and constant rising	Α
	Special take off technique required	No	Α	No	Α
2. Landing	On a stable and the set of the state of the set	Nie		N.	
3. Speed in st	Special landing technique required	No	Α	No	Α
o. opeeu iii st	Trim speed more than 30 km/h	Yes	Α	Yes	Α
	Speed range using the controls larger than 10 km/h		Α	Yes	Α
	Minimum speed	Less than 25 km/h	Α	25 km/h to 30 km/h	В
4. Control mo					
	Max. weight in flight up to 80 kg	Increasing Constanthan FF and	^	and evallable	_
	Symmetric control pressure/travel Max. weight in flight 80 kg to 100 kg	Increasing, Greater than 55 cm	Α	not available	0
	Symmetric control pressure/travel	not available	0	Increasing, 45 cm to 60 cm	С
	Max. weight in flight greater than 100 kg			3, 11 11 11	
	Symmetric control pressure/travel	not available	0	not available	0
5. Pitch stabil	ity exiting accelerated flight	P: (II II 000		D: (11 11 000	
	Dive forward angle on exit Collapse occurs		A A	Dive forward less than 30° No	A A
6 Pitch stahil	ity operating controls during accelerated flight	NO .	Α	NO .	A
o. i itoli stabii	Collapse occurs	No	Α	No	Α
7. Roll stabilit	y and damping				
	Oscillations	Reducing	Α	Reducing	Α
8. Stability in		0		On anti-manage suit	
0 Roberton:	Tendency to return to straight flight	Spontaneous exit	Α	Spontaneous exit	Α
a. Denaviour I	in a steeply banked turn Sink rate after two turns	More than 14 m/s	В	More than 14 m/s	В
10. Symmetric	c front collapse	Wore than 14 m/s		Wore than 14 m/s	
	Entry	Rocking back less than 45°	Α	Rocking back less than 45°	Α
	Recovery		Α	Spontaneous in less than 3 s	Α
	Dive forward angle on exit		В	Dive foward 0°to 30°, Keeping course	Α
	Cascade occurs	No .	Α	No	Α
	With accelerator Entry	Rocking back less than 45°	Α	Rocking back less than 45°	Α
	Recovery		A	Spontaneous in less than 3 s	A
	Dive forward angle on exit		В	Dive foward 0°to 30°, Keeping course	A
	, and the second se	90°		· • •	
	Cascade occurs	No	Α	No	Α
11. Exiting de	ep stall (parachutal stall)	Vaa	,	Van	^
	Deep stall achieved Recovery		A A	Yes Spontaneous in less than 3 s	A A
	Dive forward angle on exit	•	A	Dive forward 0°to 30°	A
	Change of course		Α	Changing course less than 45°	Α
	Cascade occurs	No .	Α	No	Α
12. High angle	e of attack recovery				
	Recovery	•	Α	Spontaneous in less than 3 s	A
13 Pecovery	Cascade occurs from a developed full stall	No	Α	No	Α
13. Recovery	Dive forward angle on exit	Dive forward 0°to 30°	Α	Dive forward 30°to 60°	В
	Collapse		Α	No collapse	A
	Cascade occurs (other than collapse)	No .	Α	No	Α
	Rocking back		Α	Less than 45°	Α
44 40	Line tension	Most line tight	Α	Most line tight	Α
14. Asymmetr					
	With 50% collapse-Maximum dive forward or roll angle Change of course until re-inflation	Less than 90°, Dive or roll angle 15° to 45°	Α	Less than 90°, Dive or roll angle 15° to 45°	Α
	Re-inflation behaviour		c	Spontaneous re-inflation	A
	Total change of course		Ā	Less than 360°	Α
	Collapse on the opposite side occurs		Α	No	Α
	Twist occurs			No	A
	Cascade occurs With 75% collapse Maximum dive forward or roll angle	No	Α	No	Α
	With 75% collapse-Maximum dive forward or roll angle Change of course until re-inflation	90° to 180°, Dive or roll angle 15° to 45°	В	90° to 180°, Dive or roll angle 15° to 45°	В
	Re-inflation behaviour	•	A	Spontaneous re-inflation	A
	Total change of course		Α	Less than 360°	A
	Collapse on the opposite side occurs			No	Α
	Twist occurs		Α	No	Α
	Cascade occurs		Α	No	Α
	With 50% collapse and accelerator-Maximum dive forward of Change of course until re-inflation		۸	Loss than 00° Dive or roll angle 15° to 45°	Λ
	Re-inflation behaviour	ing the second of the second o	A A	Less than 90°, Dive or roll angle 15° to 45° Spontaneous re-inflation	A A
	Total change of course			Less than 360°	A

	Collapse on the opposite side occurs	No	Α	No	Α
	Twist occurs	No	Α	No	Α
	Cascade occurs	No	Α	No	Α
	With 75% collapse and accelerator-Maximum dive forward o Change of course until re-inflation		В	000 to 4000 Dive or rell and a 450 to 600	С
	· · · · · · ·	90° to 180°, Dive or roll angle 15° to 45° Spontaneous re-inflation	A	90° to 180°, Dive or roll angle 45° to 60° Spontaneous re-inflation	
	Re-inflation behaviour	Less than 360°	A	Less than 360°	A A
	Total change of course				
	Collapse on the opposite side occurs	No	A	No	A
	Twist occurs Cascade occurs	No No	A	No No	A A
1E Direction	al control with a maintained asymmetric collapse	NO	А	NO	А
15. Directiona	Able to keep course	Yes	Α	Yes	Α
	180° turn away from the collapsed side possible in 10 s	Yes	A	Yes	A
	Amount of control range between turn and stall or spin	More than 50 % of the symmetric control travel	A	More than 50 % of the symmetric control travel	A
16 Trim spor	ed spin tendency	wore than 50 % of the symmetric control travel		wore than 50 % of the symmetric control travel	
10. ITIIII spee	Spin occurs	No	Α	No	Α
17. Low spee	ed spin tendency	110		110	
Low open	Spin occurs	No	Α	No	Α
18. Recovery	r from a developed spin		- / \		
. C. INCOCTORY	Spin rotation angle after release	Stops spinning in less than 90°	Α	Stops spinning in less than 90°	Α
	Cascade occurs	No	Α	No	A
19. B-line sta			,,		, ,
To. B line sta	Change of course before release	Change of course less than 45°	Α	Change of course less than 45°	Α
	Behaviour before release	Remains stable with straight span	Α	Remains stable with straight span	A
	Recovery	Spontaneous in less than 3 s	Α	Spontaneous in less than 3 s	A
	Dive forward angle on exit	Dive forward 0° to 30°	Α	Dive forward 0° to 30°	A
	Cascade occurs	No	Α	No	A
20. Big ears	Cucculo Cocuro		- / \		- / `
20. 2.9 00.0	Entry procedure	Standard technique	Α	Dedicated controls	Α
	Behaviour during big ears	Stable flight	Α	Stable flight	Α
	Recovery	Recovery through pilot action in less than a	В	Recovery through pilot action in less than a	В
	,	further 3 s		further 3 s	
	Dive forward angle on exit	Dive forward 0° to 30°	Α	Dive forward 0° to 30°	Α
21. Big ears i	in accelerated flight				
-	Entry procedure	Standard technique	Α	Dedicated controls	Α
	Behaviour during big ears	Stable flight	Α	Stable flight	Α
	Recovery	Recovery through pilot action in less than a	В	Recovery through pilot action in less than a	В
		further 3 s		further 3 s	
	Dive forward angle on exit	Dive forward 0° to 30°	Α	Dive forward 0° to 30°	Α
	Behaviour immediately after releasing the accelerator while	Stable flight	Α	Stable flight	Α
	maintaining big ears				
22. Behaviou	ır exiting a steep spiral				
	Tendency to return to straight flight	Spontaneous exit	Α	Spontaneous exit	Α
	Turn angle to recover normal flight	Less than 720°, spontaneous recovery	Α	Less than 720°,spontaneous recovery	Α
	Sink rate when evaluating spiral stability [m/s]	17 m/s		19 m/s	
23. Alternativ	ve means of directional control				
	180° turn achievable in 20 s	Yes	Α	Yes	Α
	Stall or spin occurs	No	Α	No	Α
24. Any other	r flight procedure and/or configuration described in the us				
	Procedure works as described	not available		not available	C
	Procedure suitable for novice pilots	not available	0	not available	C
	Cascade occurs	not available	0	not available	C
Comments of	f test pilot	not available	0		Ĺ
Comments of		not available	0	not available	0



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