

Harness Structural test Report

Inspection certificate number: **PH_274.2019**

Manufacturer data:

Manufacturer name: **Davinci Products Inc.**
 Representative: **Jihun Lee**
 Street: **53 sinchon-gil, Okcheon-myeon, Yangpy**
 Post code place: **Gyeonggi-do 12505**
 Country: **Republic of Korea**

Sample data:

Name: **Harmony Passenger Air**
 Type: **ABS**
 Size: **Unique**
 Serial number: **HHA-100491**
 Impact pad type: ⁽¹⁾ **Airbag**
 Clip-in weight [kg]: **100**

Date of test: **27.05.2019**

Atmosphere AGL:

[C°]	21.4
RH [%]	40
[hPa]	978.1

Summary of Structural test

Test id	- EN 1651	Setup	Req. Load [g]	Req. Load [N]	Min. duration [s]	Result
R0	✓ 5.3.2.1	Default flying position	6	6000	10	POSITIVE
R2	5.3.2.2	Default flying position	15	15000	5	n/a
R4	✓ 5.3.2.7	Flying position before landing	15	15000	5	POSITIVE
R6	5.3.2.4	Rescue attachments	15	15000	5	n/a
R8	✓ 5.3.2.3	Asymmetric, one riser	6	6000	10	POSITIVE
R9	5.3.2.5	Towing	5	5000	10	n/a
R10	✓ 5.3.2.6	Asymmetric, negative	4.5	4500	10	POSITIVE

Rescue deployment test

Test id	- NfL II 91/09	Setup	Min load [N]	Max. load [N]	Measured [N]	Result
RRDT	6.1.5	Default flying position	20	70	0.00	n/a

Rescue Deployment Handle strength test

Test id	- EN 12491	Setup	Req. Load [N]	Min. duration [s]	Breaking strength [N]	Result
RRST	5.3.2	Two end points of handle	700	10	0.00	n/a

Manufacture	Instrument	Type no	S/N	Validity Calibration
HBM	Load Sensor GE01	1-S9M/50KN-1	31314643	04.09.2023
Burster	Sensor Burster	8431-10000	1185483	04.09.2023
JDC elec	Geos n°11 Skywatch	Geos n°11	22	08.05.2020

The validation of this test report is given by the signature of the test manager on the Inspection Certificate no 94.20

⁽¹⁾ If Impact pad available, see test report no. 94.22 and inspection certificate no. 94.20

Calculated value in tests reports include the value minus the uncertainty (on safe side) / The uncertainty stated is the expanded uncertainty obtained by multiplying the standard uncertainty by the coverage factor k = 2. The value of the measurand lies within the assigned range of values with a probability of 95%.

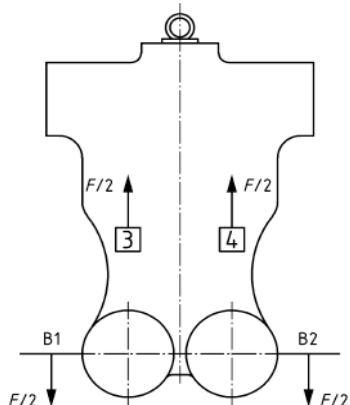
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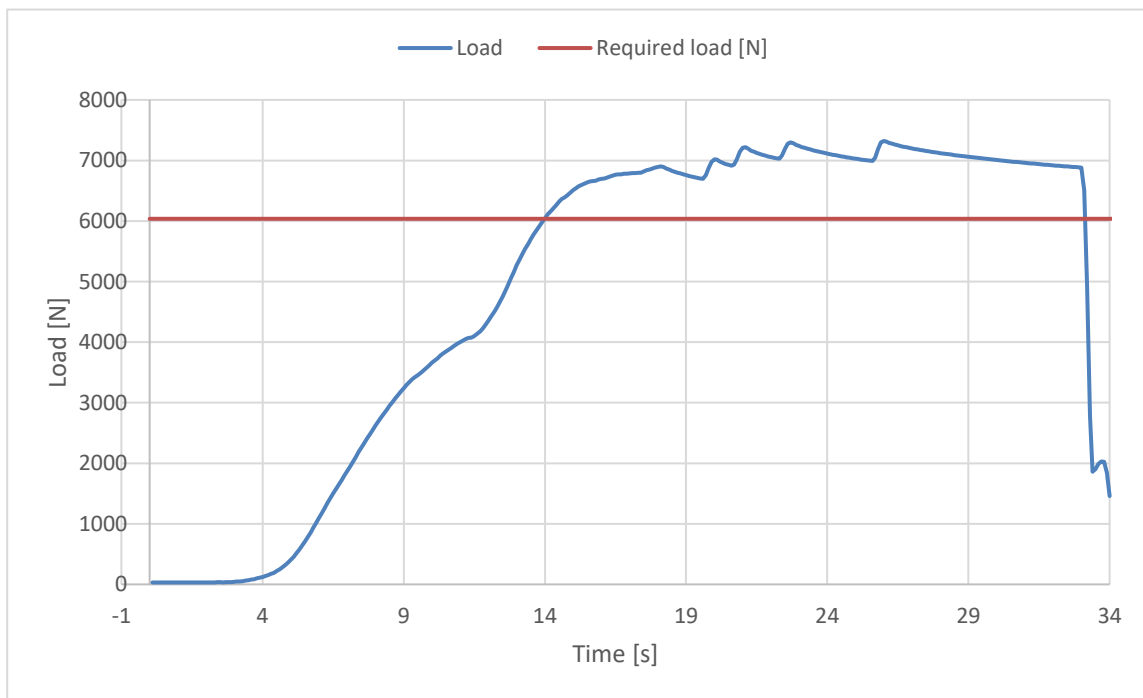
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model: **Harmony Passenger Air Unique**

Harness Structural test

Test ID R0

Standard	EN 1651:1999	
Reference in standard	5.3.2.1	
Test setup	Default flying position	
Attachment points	Both main riser attachment (3,4)	
Anchor points	Dummy (B1, B2)	
Required load [g]	6	
Required load [N]	6000	
Minimum test duration [s]	10	
Result		
Test duration [s]	19.2	
Any signs of structural failure	No	
Test results	POSITIVE	



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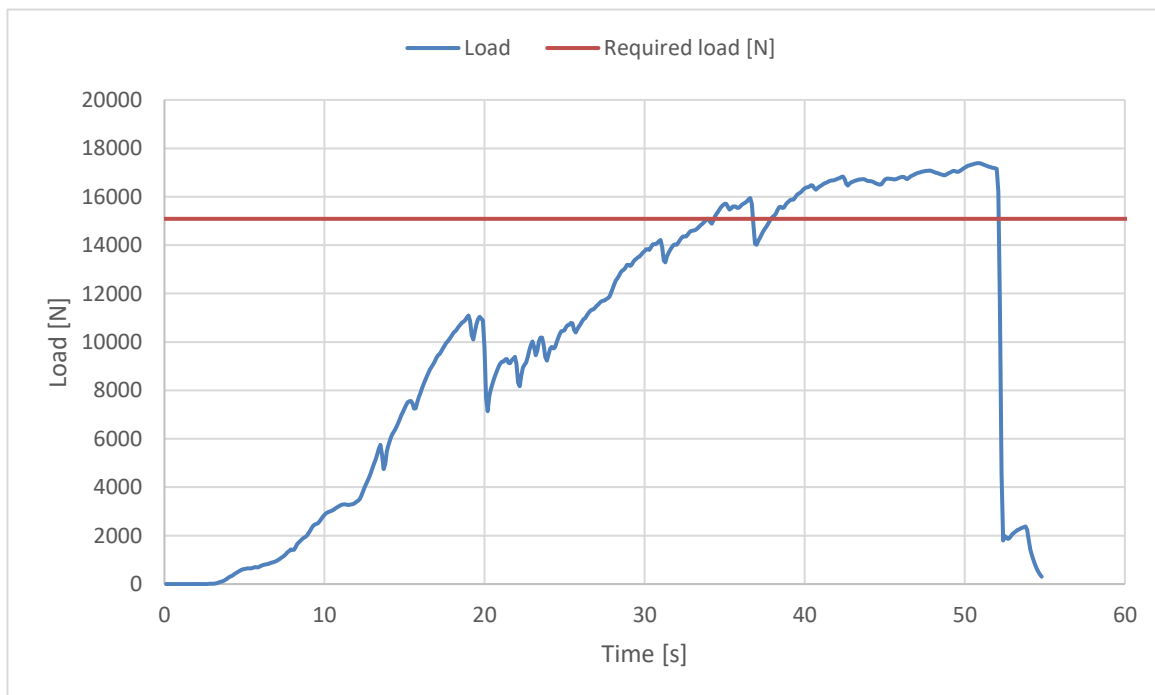
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model: **Harmony Passenger Air Unique**

Harness Structural test

Test ID R4

Standard	EN 1651:1999
Reference in standard	5.3.2.7
Test setup	Flying position before landing
Attachment points	Both main riser attachment (3,4)
Anchor points	Dummy (7,8)
Required load [g]	15
Required load [N]	15000
Minimum test duration [s]	5
Result	
Test duration [s]	14.3
Any signs of structural failure	No
Test results	POSITIVE



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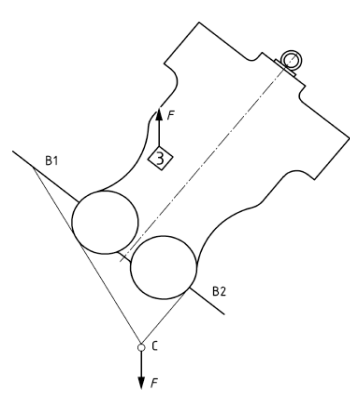
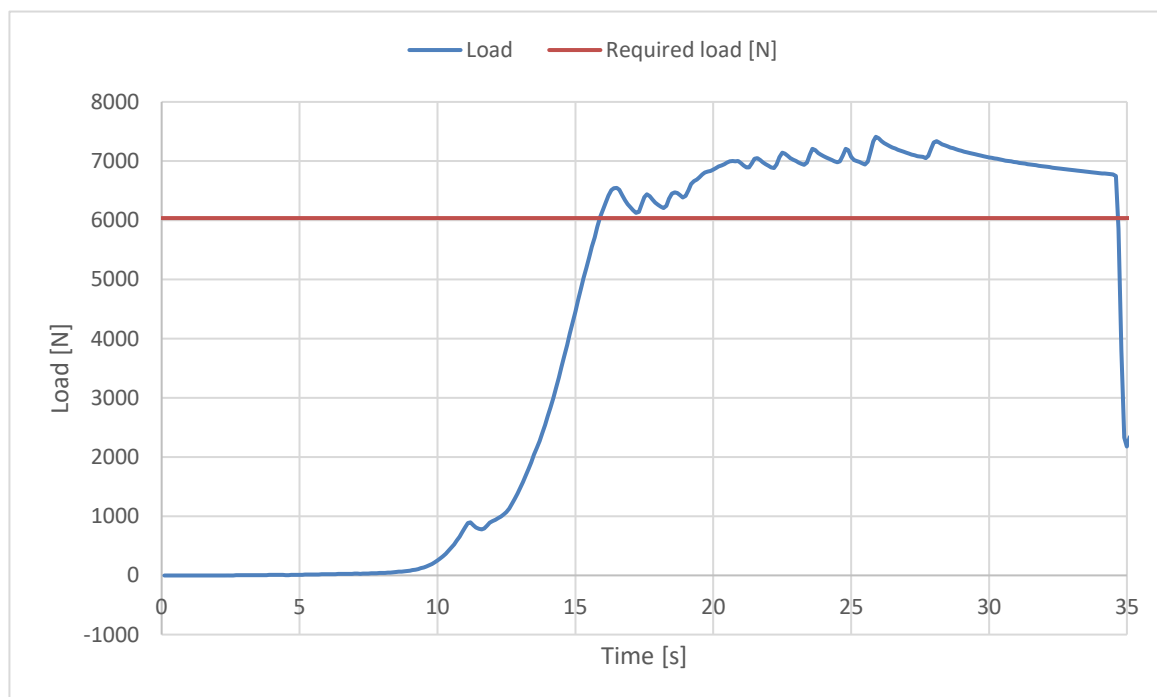
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model: **Harmony Passenger Air Unique**

Harness Structural test

Test ID R8

Standard	EN 1651:1999
Reference in standard	5.3.2.3
Test setup	Asymmetric, one riser
Attachment points	One main riser attachment (3)
Anchor points	Dummy (B1,B2)
Required load [g]	6
Required load [N]	6000
Minimum test duration [s]	10
Result	
Test duration [s]	18.8
Any signs of structural failure	No
Test results	POSITIVE

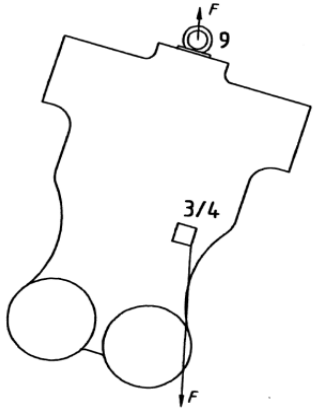
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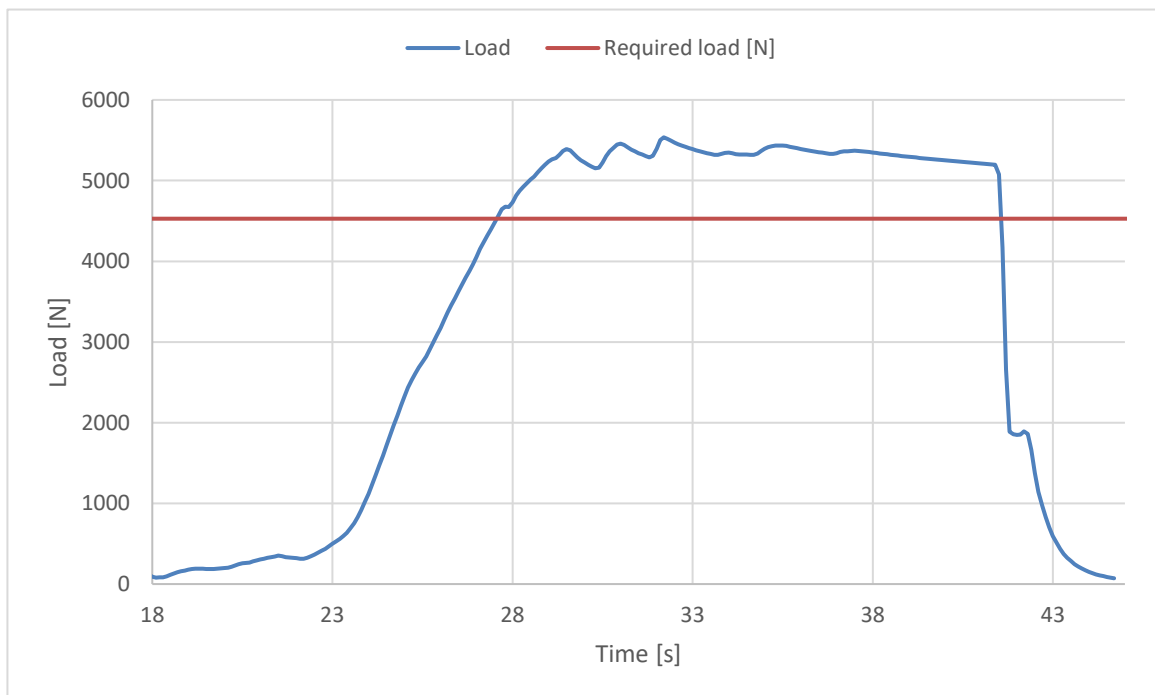
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Harness Structural test

Test ID R10

Standard	EN 1651:1999	
Reference in standard	5.3.2.6	
Test setup	Asymmetric, negative	
Attachment points	One main riser attachment (3 or 4) downwards	
Anchor points	Dummy (9)	
Required load [g]	4.5	
Required load [N]	4500	
Minimum test duration [s]	10	
Result		
Test duration [s]	14	
Any signs of structural failure	No	
Test results	POSITIVE	



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