Test laboratory for paragliders, paraglider harnesses and paraglider reserve parachutes



Paragliders Shock- and sustained loading test

Inspection certificat number:	PG_1374.2018		Test Report
Manufacturer data			
Manufacturer name:	Sky Paragliders		
Representative:	Nemec Martin		
Street:	Okruzni 39		
Post code / place:	73911 Frydlant N.C.		
Country:	Czech Republic		
Sample data			
Name:	Swift 5		
Size:	L		
Maximum weight in flight [kg]:	115		
Serial number:	PR2-T-35A-005		
Date of reception:	06.08.2018		
Test data		Test Atmosphere AGL	
Place of test:	Yverdon (airport)	5	[°C]
Date of test:	27.09.2018	73	RH [%]
Inspector:	Alain Zoller	979.9	[hPA]
		0.1	Wind [m/s]

Shock loading test result ⁽¹⁾			
Weak link used [daN]: Visual inspection:	1000 No visible damage	Results:	POSITIVE
Uncertainty k=2 [%] (2)	10		

Weak link



Instruments	Validity	Manufacturer	s/n
Weak link	2020	Tost	n/a
Cable	2020	Rotex	n/a
Geos nº 11 Skywatch	08.05.2019	JDC elec.	22

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Sustained loading test results ⁽³⁾

Result :	POSITIVE
Calculated max load value with 3 sec or five peaks [kg] :	123.83

Required sustained loading test results⁽⁴⁾

Required load value for 3s at 8g [N] :	9025.20
Required load value for 5 peaks at 10g [N] :	11281.50
Required load value for 3s at 8g includes uncertainty [N] :	9087.96
Required load value for 5 peaks at 10g includes uncertainty [N] :	11344.26
Uncertainty K=2 [%] :	0.487

Graphic sustained loading diagram



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Inspection certificate number:	PG_1374.2018		
Detailed sustained loading test results			
Calculated cumulative duration at m	ax load [s] :	3.6	
Calculated max load value duration	of 3 sec. [N] :	1214.78	
Calculated max load value duration	of 3 sec. [kg] :	123.83	
Calculated max load value with five	peaks [N] :	n/a	
Calculated max load value with five	peaks [kg] :	n/a	
Calculated max load value with 3 se	c or five peaks [N] :	1214.78	
Calculated max load value with 3 se	c or five peaks [kg] :	123.83	

Instruments	Manufacturer	Type nr.	S/N
Load sensor	НВМ	1-S9M/50KN-1	31314652
Geos n°11 Skywatch	JDC	Geos nº 11	0022

The validation of this test report is given by the signature of the test manager on inspection certificate 71.8.1

Air Turquoise SA has thoroughly tested the sample of paraglider mentioned above and certifies its conformity with the standards EN 926-1:2015 chapter 4.4, 4.5 | LTF NFL II-91/09 chapter 3

(1) The paraglider is subjected to a shock load . Shock load is limited using a weak link according to the weight range of glider. The weak link breaks or 5 s has elapsed since the start of the shock load. The wing is then visually inspected for damage.

(2) Weak link value include the uncertainty for weight range test values / The uncertainty state is the expanded uncertainty obtained by multiplying the standard uncertainty by the coverage factor k = 2. The value of the measurand lies within the assigned range of values with a probability of 95%.

(3) The test specimen (sample) is attached to the electronic sensors on the tow vehicle.

A controller is positioned on the tow vehicle in order to operate the paraglider control lines to stabilize the wing. The speed of the vehicle is increased as gradually as possible, enabling the controller to obtain satisfactory stabilisation of the flight path of the paraglider.

When the paraglider has stabilized, the speed is increased gradually until either:

a) the measured load exceeds a load factor of eight times the maximum total weight in flight recommended by the manufacturer, for a minimum cumulative duration of 3 s; or

b) five peaks separated by at least 0,3 s are obtained above ten times the maximum total weight in flight recommended by the manufacturer, in one run.

(4) The calculated value include the value minus the uncertainty / The uncertainty stated is the expanded uncertainty obtained by multiplying the standard uncertainty by the coverage factor k = 2. The value of the measurand lies within the assigned range of values with a probability of 95%.