## Flight test report

Manufacturer ADVANCE Thun AG

Address Seestrasse 14 3602 Thun Switzerland

Representive Eisenhut Kari Type of glider Beta 4 42 Open trimmer Trimmer

PG 128.2008 Certification number Date of flight test 08/02/2008 Villeneuve Place of test

Alain Zoller



## Classification C

Test Pilot Claude Thurnheer Harness Advance Bi-pro 50 cm

Advance - Bi Pro 2 225 kg Total weight in flight 135 kg

		Min weight		Max weight	
1. Inflation/Ta		wiiii weignit		max weight	
	Rising behaviour	Smooth, easy and constant rising	Α	Smooth, easy and constant rising	Α
	Special take off technique required		Α	No	Α
2. Landing					
	Special landing technique required	No	Α	No	Α
3. Speed in st		V	٨	V	
	Trim speed more than 30 km/h		A	Yes	A
	Speed range using the controls larger than 10 km/h Minimum speed		A B	Yes Less than 25 km/h	A A
4. Control mo		25 KIIVII IO 50 KIIVII	Б	Less than 25 km/n	
4. 001111011110	Max. weight in flight up to 80 kg				
	Symmetric control pressure/travel	not available	0	not available	0
	Max. weight in flight 80 kg to 100 kg				
	Symmetric control pressure/travel	not available	0	not available	0
	Max. weight in flight greater than 100 kg				
E Ditab atabil	Symmetric control pressure/travel	Increasing, Greater than 65 cm	Α	Increasing, Greater than 65 cm	Α
5. Pitch Stabil	lity exiting accelerated flight Dive forward angle on exit	not available	0	not available	0
	Collapse occurs	not available		not available	0
6. Pitch stabil	lity operating controls during accelerated flight	not available	Ü	Tiot dvalidate	Ŭ
	Collapse occurs	not available	0	not available	0
7. Roll stabilit	ty and damping				
	Oscillations	Reducing	Α	Reducing	Α
8. Stability in					
O Deberrie	Tendency to return to straight flight	Spontaneous exit	Α	Spontaneous exit	Α
9. Benaviour	in a steeply banked turn Sink rate after two turns	More than 14 m/s	В	More than 14 m/s	В
10 Symmetri	c front collapse	More than 14 m/s	ь	More than 14 m/s	ь
To. Symmetri	Entry	Rocking back less than 45°	Α	Rocking back less than 45°	Α
	Recovery		В	Spontaneous in less than 3 s	Α
	Dive forward angle on exit		Α	Dive foward 0°to 30°, Keeping course	Α
		90°			
	Cascade occurs	No	Α	No	Α
	With accelerator				_
	Entry	not available		not available	0
	Recovery Dive forward angle on exit	not available not available	0	not available not available	0
	Cascade occurs	not available	_	not available	0
11. Exiting de	ep stall (parachutal stall)				
ŭ	Deep stall achieved	Yes	Α	Yes	Α
	Recovery	Spontaneous in less than 3 s	Α	Spontaneous in less than 3 s	Α
	Dive forward angle on exit		Α	Dive forward 0°to 30°	Α
	Change of course	0 0	Α	Changing course less than 45°	Α
40 High and	Cascade occurs	No	Α	No	Α
12. High angi	e of attack recovery Recovery	not available	0	not available	0
	Cascade occurs	not available	0	not available	0
13. Recovery	from a developed full stall		-		
	Dive forward angle on exit	Dive forward 30°to 60°	В	Dive forward 30°to 60°	В
	Collapse	No collapse	Α	No collapse	Α
	Cascade occurs (other than collapse)			No	Α
	Rocking back		Α	Less than 45°	Α
14 Agreement	Line tension	Most line tight	Α	Most line tight	Α
14. Asymmet					
	With 50% collapse-Maximum dive forward or roll angle Change of course until re-inflation	Less than 90°, Dive or roll angle 15° to 45°	Α	Less than 90°, Dive or roll angle 0° to 15°	Α
	Re-inflation behaviour		A	Spontaneous re-inflation	A
	Total change of course		Α	Less than 360°	Α
	Collapse on the opposite side occurs		Α	No	Α
	Twist occurs	No	Α	No	Α
	Cascade occurs	No	Α	No	Α
	With 75% collapse-Maximum dive forward or roll angle				
	Change of course until re-inflation			not available	0
	Re-inflation behaviour			not available	0
	Total change of course			not available not available	0
	Collapse on the opposite side occurs Twist occurs			not available	0
	Cascade occurs			not available	0
	With 50% collapse and accelerator-Maximum dive forward or		•		J
	Change of course until re-inflation	not available	0	not available	0
	Re-inflation behaviour	not available	0	not available	0
	Total change of course	not available	0	not available	0

	Collapse on the opposite side occurs	not available	0	not available	0
	Twist occurs	not available	0	not available	0
	Cascade occurs	not available	0	not available	0
	With 75% collapse and accelerator-Maximum dive forward o				
	Change of course until re-inflation	not available	0		0
	Re-inflation behaviour	not available	0	not available	0
	Total change of course	not available	0	not available	0
	Collapse on the opposite side occurs	not available	0	not available	0
	Twist occurs	not available	0	not available	0
	Cascade occurs	not available	0	not available	0
15. Directiona	al control with a maintained asymmetric collapse				
	Able to keep course	Yes	Α	Yes	Α
	180° turn away from the collapsed side possible in 10 s	Yes	Α	Yes	Α
	Amount of control range between turn and stall or spin	More than 50 % of the symmetric control travel	Α	More than 50 % of the symmetric control travel	Α
16. Trim spee	d spin tendency				
	Spin occurs	No	Α	No	Α
17. Low spee	d spin tendency				
	Spin occurs	No	Α	No	Α
18. Recovery	from a developed spin				
	Spin rotation angle after release	Stops spinning in less than 90°	Α	not available	0
	Cascade occurs	No	Α	not available	0
19. B-line stal	I				
	Change of course before release	not available	0	not available	0
	Behaviour before release	not available	0	not available	0
	Recovery	not available	0	not available	0
	Dive forward angle on exit	not available	0	not available	0
	Cascade occurs	not available	0	not available	0
20. Big ears					
_	Entry procedure	Dedicated controls	Α	Dedicated controls	Α
	Behaviour during big ears	Stable flight	Α	Stable flight	Α
	Recovery	Recovery through pilot action in less than a	В	Spontaneous in 3 s to 5 s	В
		further 3 s			
	Dive forward angle on exit	Dive forward 0° to 30°	Α	Dive forward 0° to 30°	Α
21. Big ears in	n accelerated flight				
_	Entry procedure	not available	0	not available	0
	Behaviour during big ears	not available	0	not available	0
	Recovery	not available	0	not available	0
	Dive forward angle on exit	not available	0	not available	0
	Behaviour immediately after releasing the accelerator while	not available	0	not available	0
	maintaining big ears				
22. Behaviou	r exiting a steep spiral				
	Tendency to return to straight flight	Spontaneous exit	Α	Spontaneous exit	Α
	Turn angle to recover normal flight	•	Α	Less than 720°, spontaneous recovery	Α
	Sink rate when evaluating spiral stability [m/s]	23 m/s		25 m/s	
23. Alternativ	e means of directional control				
	180° turn achievable in 20 s	Yes	Α	Yes	Α
	Stall or spin occurs		A	No	Α
24. Any other	flight procedure and/or configuration described in the us				
,	Procedure works as described	not available	0	not available	0
	Procedure suitable for novice pilots	not available	0	not available	0
	Cascade occurs	not available	0	not available	0
Comments of		not available	U	TIOL GYGIIGDIO	U
Comments of	Comments	no		no	



Air Turquoise
Rue de la Poterlaz 6
Case postale 10
CH- 1844 Villeneuve
Switzerland
mobile: +41 79 202 52 30
Tel. no: +41 21 965 65 65
fax: +41 219 65 65 66
email: info@airturquoise.ch
homepage: www.para-test.com



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