Flight test report

Manufacturer Gradient s.r.o. Plzeňská 221/130 Address

150 00 Praha 5 - Motol

Czech Republic

Representive None Type of glider Bright III 24 not available Trimmer

PG 025.2006 Certification number Date of flight test 13/03/2007 Villeneuve Place of test



Classification B

Test Pilot Philippe Dupont Harness Sup'Air Light Claude Thurnheer sup air light 80 kg Total weight in flight 60 kg

		Min weight	Max weight
1. Inflation/Ta	ke-off		
	Rising behaviour Special take off technique required	Smooth, easy and constant rising A No A	
2. Landing	Special take on technique required	NO A	, NO
	Special landing technique required	No A	No /
3. Speed in st			
	Trim speed more than 30 km/h Speed range using the controls larger than 10 km/h	Yes A Yes A	
	Minimum speed	Less than 25 km/h	
4. Control mo		Less trail 25 km/m	LCGS (Hall 25 KHI/H
	Max. weight in flight up to 80 kg		
	Symmetric control pressure/travel	Increasing, Greater than 55 cm	not available
	Max. weight in flight 80 kg to 100 kg Symmetric control pressure/travel	not available	Increasing, Greater than 55 cm
	Max. weight in flight greater than 100 kg	TIOL AVAIIABLE	Increasing, Greater than 55 cm
	Symmetric control pressure/travel	not available	not available
Pitch stabil	ity exiting accelerated flight		
	Dive forward angle on exit	Dive forward less than 30° A	
6 Ditch stabil	Collapse occurs ity operating controls during accelerated flight	No A	No /
o. Fitch stabil	Collapse occurs	No A	No
7. Roll stabilit	y and damping		
	Oscillations	Reducing A	Reducing
8. Stability in	gentle spirals Tendency to return to straight flight	Spontonogua ovit	Spontaneous exit
9 Rehaviour i	in a steeply banked turn	Spontaneous exit A	Spontaneous exit
J. Bellaviour	Sink rate after two turns	12 m/s to 14 m/s	More than 14 m/s
10. Symmetric	c front collapse		
	Entry	Rocking back less than 45° A	The second secon
	Recovery	Spontaneous in less than 3 s	
	Dive forward angle on exit Cascade occurs	Dive foward 0°to 30°, Keeping course A No A	·
	With accelerator		,
	Entry	Rocking back less than 45° A	Rocking back less than 45°
	Recovery	Spontaneous in less than 3 s	
	Dive forward angle on exit	Dive foward 0°to 30°, Keeping course A	
11 Exiting de	Cascade occurs ep stall (parachutal stall)	No A	No /
TT. Exiting uc	Deep stall achieved	Yes	Yes
	Recovery	Spontaneous in less than 3 s	Spontaneous in less than 3 s
	Dive forward angle on exit	Dive forward 0°to 30°	
	Change of course Cascade occurs	Changing course less than 45° A No A	
12. High angle	e of attack recovery	NO A	1 140
g ug	Recovery	Spontaneous in less than 3 s	Spontaneous in less than 3 s
	Cascade occurs	No A	No /
13. Recovery	from a developed full stall	D: 6 1001 000	B: 6 1000 000
	Dive forward angle on exit Collapse	Dive forward 0°to 30° No collapse A	
	Cascade occurs (other than collapse)	No A	•
	Rocking back	Less than 45° A	
	Line tension	Most line tight	Most line tight
14. Asymmetr			
	With 50% collapse-Maximum dive forward or roll angle Change of course until re-inflation	Less than 90°, Dive or roll angle 0° to 15°	Less than 90°, Dive or roll angle 0° to 15°
	Re-inflation behaviour	Spontaneous re-inflation A	
	Total change of course	Less than 360° A	
	Collapse on the opposite side occurs	No A	
	Twist occurs		No No
	Cascade occurs With 75% collapse-Maximum dive forward or roll angle	No A	No /
	Change of course until re-inflation	Less than 90°, Dive or roll angle 15° to 45°	Less than 90°, Dive or roll angle 15° to 45°
	Re-inflation behaviour	Spontaneous re-inflation A	
	Total change of course	Less than 360° A	
	Collapse on the opposite side occurs	No A	
	Twist occurs Cascade occurs	No A	
	With 50% collapse and accelerator-Maximum dive forward or		,
	Change of course until re-inflation	90° to 180°, Dive or roll angle 0° to 15° A	Less than 90°, Dive or roll angle 0° to 15°
	Re-inflation behaviour	Spontaneous re-inflation A	Spontaneous re-inflation
	Total change of course	Less than 360° A	
	Collapse on the opposite side occurs	No A	No A

	Twist occurs	No		No	Α	
	Cascade occurs	No	Α	No	Α	
	With 75% collapse and accelerator-Maximum dive forward of					
	Change of course until re-inflation	90° to 180°, Dive or roll angle 0° to 15°	Α	Less than 90°, Dive or roll angle 15° to 45°	Α	
	Re-inflation behaviour	Spontaneous re-inflation	Α	Spontaneous re-inflation	Α	
	Total change of course	Less than 360°	Α	Less than 360°	Α	
	Collapse on the opposite side occurs	No	Α	No	Α	
	Twist occurs	No	Α	No	Α	
	Cascade occurs	No	Α	No	Α	
15. Direction	al control with a maintained asymmetric collapse					
	Able to keep course	Yes	Α	Yes	Α	
	180° turn away from the collapsed side possible in 10 s	Yes	Α	Yes	Α	
	Amount of control range between turn and stall or spin	More than 50 % of the symmetric control travel	Α	More than 50 % of the symmetric control travel	Α	
16. Trim spec	ed spin tendency	, , , , , , , , , , , , , , , , , , ,		,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,		
. от т.т оро	Spin occurs	No	Α	No	Α	
17. Low spec	ed spin tendency				- '	
_ opec	Spin occurs	No	Α	No	Α	
18. Recovery	r from a developed spin		- ' '		,,	
To. Recovery	Spin rotation angle after release	Stops spinning in less than 90°	Α	Stops spinning in less than 90°	Α	
	Cascade occurs	No	A	No	A	
19. B-line sta		NO	А	INO	А	
19. D-IIIle Sta	Change of course before release	Change of course less than 45°	Α	Change of course less than 45°	Α	
	Behaviour before release		A	Remains stable with straight span	A	
		Remains stable with straight span				
	Recovery	Spontaneous in less than 3 s	Α	Spontaneous in less than 3 s	Α	
	Dive forward angle on exit	Dive forward 0° to 30°	Α	Dive forward 0° to 30°	Α	
	Cascade occurs	No	Α	No	Α	
20. Big ears						
	Entry procedure	Dedicated controls	Α	Dedicated controls	Α	
	Behaviour during big ears	Stable flight	Α	Stable flight	Α	
	Recovery	Spontaneous in less than 3 s	Α	Spontaneous in less than 3 s	Α	
	Dive forward angle on exit	Dive forward 0° to 30°	Α	Dive forward 0° to 30°	Α	
21. Big ears in accelerated flight						
	Entry procedure	Dedicated controls	Α	Dedicated controls	Α	
	Behaviour during big ears	Stable flight	Α	Stable flight	Α	
	Recovery	Spontaneous in less than 3 s	Α	Spontaneous in less than 3 s	Α	
	Dive forward angle on exit	Dive forward 0° to 30°	Α	Dive forward 0° to 30°	Α	
	Behaviour immediately after releasing the accelerator while	Stable flight	Α	Stable flight	Α	
22. Behaviou	r exiting a steep spiral					
	Tendency to return to straight flight	Spontaneous exit	Α	Spontaneous exit	Α	
	Turn angle to recover normal flight	Less than 720°, spontaneous recovery	Α	Less than 720°,spontaneous recovery	Α	
	Sink rate when evaluating spiral stability [m/s]	17 m/s		19 m/s		
23. Alternativ	ve means of directional control					
	180° turn achievable in 20 s	Yes	Α	Yes	Α	
	Stall or spin occurs	No	Α	No	Α	
24. Any othe	r flight procedure and/or configuration described in the us					
, out	Procedure works as described	not available	0	not available	0	
	Procedure suitable for novice pilots	not available	0	not available	0	
	Cascade occurs	not available	0	not available	0	
Comments o		HOL AVAIIADIC	U	TIOL AVAIIADIC	U	
Comments 0	Comments	no		no		
	Continuents	110		110		



Air Turquoise
Rue de la Poterlaz 6
Case postale 10
CH- 1844 Villeneuve
Switzerland
mobile: +41 79 202 52 30
Tel. no: +41 21 965 65 65
fax: +41 219 65 65 66
email: info@airturquoise.ch
homepage: www.cen.li